Statement A: Approved for public release. Distribution is unlimited.



Yale Chang

Considerations for Implementing Presidential Memorandum-20 Guidelines for Nuclear Safety Launch Authorization for Future Civil Space Missions



Nuclear Technology Journal Paper



CONSIDERATIONS FOR IMPLEMENTING PRESIDENTIAL MEMORANDUM-20 GUIDELINES FOR NUCLEAR SAFETY LAUNCH AUTHORIZATION FOR FUTURE CIVIL SPACE MISSIONS

The Johns Hopkins University Applied Physics Laboratory, 11199 Johns Hopkins Rd, Laurel, MD 20725

National Security Presidential Memorandum-20 (NSPM-20) (Launch of Spacecraft Containing Space Nuclear Systems) [1] dated 20 August 2019 provides updated guidelines for branch authorization for three categories of proposed launches of spacecraft with space nuclear systems: Federal Government civil space including NASA, Federal Government defense and stelligence, and commercial. These space nuclear systems provide power, heat, and/or propulsion to the

NSPM-20 status: "For United States launches of space nuclear systems, the Federal Government must ensure a rigorous, risk informed safety analysis and launch authorization process" [1], primarily by examining the probabilities of potential leanth and mentry accidents and their consequences. At the same time, for previous NASA missions, the Issueh approval process "has taken an assumes of six years and costs over \$40 million" [2]. In an effort to streamline the process, and improve cost and schedule, NSPM-20 provides specific guidelines including the following: (1) "to the scener possible, safety analyses and reviews should incorporate previous mission and review experience" [a.g., Environmental Impact Statements (EISs), Records of Decision (RODs), Safety Analysis Reports (SARs), and Sufety Evaluation Reports safety basis envelope established in the system-specific SAR, in which case it is not necessary to report the analysis supporting the system-specific SAR," and (3) "authorization for launches of smoothly containing space nuclear systems shall follow a three-tiered process based on the characteristics of the system, the level of potential becard, and national security considerations" (i.e., use risk-adjusted metrics for required level of effort and leunch authorization authority).

A fixture example interplanetary mission (EIM) that plans to use a Multi-Mission Radioisotope Thermoelectric Generator (MMRTG) is covered by NSPM-20, and is used here as a provey to illustrate potential considerations for implementing NSPM-20 guidelines. Assume that this EIM plans to use one or two Earth County Assist (EGA) maneuvers in its mission trajectory and that it will use launch vehicle (I.V) stages with solid propellant. In I.V This paper investigates each of these three NSPM-20 guidelines for three accident categories associated with the EIM: (1) EGA (aka Earth flyby or Earth swing-by) reentry, (2) solid propellant fires, and (3) PTS functions and probabilities. This paper also identifies the components needed to implement each guideline in a rigorous fashion, then assesses whether the necessary components (e.g., analyses, reports, tests, reviews, risk communications, previous launch approvals) currently exist or would need to be produced or modified.

Although these NSPM-20 guidelines could be logical and appropriate approaches for evaluating the risk associated with a system-of-systems (i.e., launch of nuclear protects) that has reached steady-state, the current state of affairs for the EIM is likely still in the "start-up transient" phase. For example, past EISs and SARs for each successive mission were constantly undated with new test data, new technology, new knowledge, and new understanding, such that provious risk results could change [3]. Additionally, part SARs proactively considered review comments and findings from part SERs. Therefore, one potential future side effect of the proposed cost improvement approach is the stagnation of technological progress in nuclear safety analyses.

Because there are no cases of leanth authorization of unclassified detailed guidelines for launch authorization of defense or intelligence launches of nuclear systems, this paper refers to relevant NASA missions.

The three specific NSPM-20 guidelines for the three EIM accident categories are discussed next.

LA. Earth Gravity Assist Reentry Accident

The last NASA radioisotope power systems (RPS) missions with EGA manurers were Galileo (1989) launch) and Causini (1997 launch). Galileo successfully flow a Venue-Earth-Earth-Gravity-Assist (VEEGA) mission trajectory, and Cassini successfully flow a Venus-Venus-Earth-Juniter-Gravity-Assist (VVEXIA) mission

1.4.1 Province Minster Anchors and Bestey Provinces

https://nets2020.ornl.gov/wpcontent/uploads/2020/09/TRACK-3-Full-submission.pdf

NUCLEAR TECHNOLOGY



https://www.ans.org/pubs/iournals/nt/

© 2021 American Nuclear Society DOI: https://doi.org/10.1080/00295450.2020.1855946

Technical Note

@ANS

Considerations for Implementing Presidential Memorandum-20 Guidelines for Nuclear Safety Launch Authorization for Future Civil Space Missions

Yale Chang*

The Johns Hopkins University, Applied Physics Laboratory, Laurel, Maryland 20723

Accepted for Publication November 23, 2020

Abstract - National Security Presidential Memorandum-20 (NSPM-20) (Launch of Spacecraft Containing Space Nuclear Systems) provides updated guidelines for launch authorization for three categories of proposed launches of spacecraft with space nuclear systems: Federal Government civil space including the National Aeronautics and Space Administration (NASA), Federal Government defense and intelligence, and commercial. These space nuclear systems provide power, heat, and/or propulsion to the spacecraft. NSPM-20 requires a rigorous, risk-informed safety analysis and launch authorization process. The launch approval process of previous NASA missions each took several years and cost multimillion dollars. NSPM-20 provides guidelines to potentially streamline the process and improve cost and schedule. This technical note examines three NSPM-20 guidelines on a future example interplanetary mission (EIM) as to their potential implementation feasibility for three accident categories: Earth gravity assist reentry, solid propellant fires, and flight termination system design. It is found that the safety technology for these accidents were constantly improved over the last several missions, but in some cases may not be adequate for direct use in the EIM's launch authorization process.

Keywords - Accident categories, cost improvements, NSPM-20, nuclear safety.

National Security Presidential Memorandum-20 (NSPM-20; Ref. 1), dated August 20, 2019, provides consequences. At the same time, for previous NASA misupdated guidelines for launch authorization for three categories of proposed launches of spacecraft with space guidelines "has taken an average of six years and costs muclear systems: Federal Government civil space including over 40 USD million 3 In an effort to streamline the the National Aeronautics and Space Administration (NASA), Federal Government defense and intelligence, and commercial. These space nuclear systems provide power, heat, and/or propulsion to the spacecraft. NSPM-20 replaces the ninth numbered paragraph in Presidential Directive/National Security Council-25 (PD/NSC-25; Ref. 2), originally issued December 14, 1977, pertaining to

a rigorous, risk-informed safety analysis and launch authorization process," primarily by examining the probabilities of potential launch and reentry accidents and their sions, the launch approval process under PD/NSC-25 process and to improve cost and schedule, NSPM-20 provides specific guidelines including the following: (1) "to the extent possible, safety analyses and reviews should incorporate previous mission and review experience fe.g., Environmental Impact Statements (EISs), Records of Decision (RODs), Safety Analysis Reports (SARs), and Safety Evaluation Reports (SERs)], (2) "demonstrate that the mission is within the safety basis envelope estab-NSPM-20 states: "For United States launches of space lished in the system-specific SAR, in which case it is not

https://www.tandfonline.com/eprint/WPBSFBZHDI2ICQT6 A9CM/full?target=10.1080/00295450.2020.1855946

About Me

- Yale Chang
- Johns Hopkins University Applied Physics Laboratory (JHU/APL)
- Yale.Chang@jhuapl.edu
- Experience
 - PM, Aerospace Nuclear Safety Program
 - PI, Solid Propellant Fires Characterization Pgm
 - Deputy Mgr, National Environmental Policy Act compliance / Launch Approval for New Horizons
 - APL Payload Rep, New Horizons launch



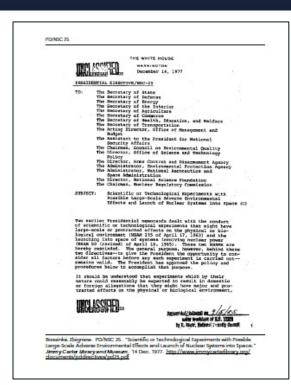
APL Payload Representative, New Horizons launch, Radiological Control Center (RADCC) at the NASA Kennedy Space Center, FL, January 2006

Launch of Space Nuclear Systems

 Any scientific or technological experiment either licensed or funded by Federal Agencies, which could reasonably be expected to result in major and protracted effects on the physical or biological environment, requires an environmental impact statement and Presidential launch approval.

Presidential Directive / National Security Council Memorandum 25 (PD/NSC 25), December 14, 1977

- Covered NASA missions Galileo, Cassini, New Horizons, Mars Science Laboratory, and Mars 2020 with Radioisotope Thermoelectric Generators (RTGs)
- Complex process viewed by some as costly and lengthy, with few specific requirements or guidelines



National Security Presidential Memorandum-20 (NSPM-20) (1/3)

- **NSPM-20** (Launch of Spacecraft Containing Space Nuclear Systems [that provide power, heat, and/or propulsion]), August 20, 2019, provides updated guidelines for launch authorization of:
 - · Commercial,
 - · Federal Government defense and intelligence,
 - Federal Government civil space including NASA
- NSPM-20 states "For United States launches of space nuclear systems, the Federal Government must ensure a rigorous, risk informed safety analysis and launch authorization process" primarily by examining the probabilities of potential launch and reentry accidents and their consequences.
- At the same time, for previous NASA missions, the launch approval process "has taken an average of six years and costs over \$40 million." [Howieson, Behrens, Kowal, NETS 2019]



NSPM-20 (2/3)

- In an effort to streamline the process, and reduce cost and schedule, NSPM-20 provides specific **guidelines**:
 - 1 "To the extent possible, safety analyses and reviews should incorporate **previous mission [analysis] and review experience**" (e.g., Environmental Impact Statements (EISs), Records of Decision (RODs), Safety Analysis Reports (SARs), and Safety Evaluation Reports (SERs);
 - 2 "demonstrate that the mission is within the safety basis envelope established in the **system-specific SAR**, in which case it is not necessary to repeat the analysis supporting the system-specific SAR"; and
 - 3 "Authorization for launches of spacecraft containing space nuclear systems shall follow a **three-tiered process based upon** the characteristics of the system, the level of potential hazard, and national security considerations," i.e., use **risk-adjusted metrics** for required level of effort (LOE) and launch authorization authority.

NSPM-20 (3/3)

- A future example interplanetary mission (EIM) that plans to use a Multi-Mission Radioisotope Thermoelectric Generator (MMRTG) will be covered by NSPM-20. Assume that this EIM
 - I.A plans to use one or two **Earth Gravity Assist (EGA)** maneuvers in its mission trajectory,
 - I.B will use launch vehicle (LV) stages with solid propellant,
 - I.C its LV will definitely have a **flight termination system (FTS)**.

These are the EIM's three accident categories

Comparison of Cost & Schedule for NSPM-20 relative to PD/NSC-25 for EIM



The **Objectives** are to **Improve** Cost & Schedule via these **Guidelines**.

	NSPM-20 Guidelines			
Accident Category	Previous analyses & reviews	System-specific SAR	Tier I, II, or III	
Earth Gravity Assist Reentry	Improve	Improve	Improve	
Solid Propellant Fires	Improve	Improve	Improve	
Flight Termination System	Improve	Improve	Improve	

8

Comparison of Cost & Schedule for NSPM-20 relative to PD/NSC-25 for EIM

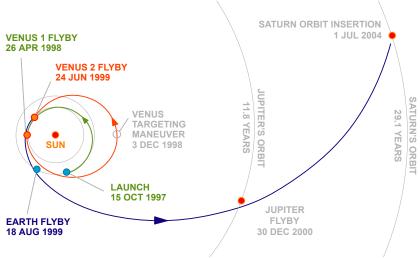


Realistically, Cost & Schedule will be Similar to past experiences.

	NSPM-20 Guidelines		
Accident Category	Previous analyses & reviews	System-specific SAR	Tier I, II, or III
Earth Gravity Assist Reentry	INSRP disagreement w/FSAR; earlier technology (25-50 yr)	Environments exceed ground test capabilities	SER estimates may trigger Tier III
Solid Propellant Fires	Different EIS risks; different test results at SL and Hi Alt	Measured environments exceed MPs of Pu and Ir	MSL EIS exceeds Tier III thresholds
Flight Termination System	Risk estimates sensitive to assumptions and timing	Different FTSs for different LVs	FTS tuned for safety vs execution

9

Earth Gravity Assist Reentry Accident



Cassini VVEJGA mission trajectory

Previous analyses & reviews

Cassini's Interagency Nuclear Safety Review Panel (INSRP) found the Safety Analysis Report had a surface energy balance error, bias in heat transfer rates, underestimation of aerodynamic heating uncertainties, and unrealistic reentry orientation. Also GPHS module structural changes (Step 0 vs Step 2 designs).

System-specific SAR

Predicted environments at 19.5 km/s (1000 g's, >7500° F, 400 MW/m²) exceed ground-based test capabilities, such as arc jet tunnels

Tier I, II, or III launch authorization

INSRP: "hypothetical 'collective cancer risk' of about 1500 latent cancer fatalities"
Possibly Tier III (Presidential)

Solid Propellant Fire Accident



Mid-Atlantic Regional Spaceport Pad 0A is seen on Oct. 28, 2014 after Orbital Sciences' Antares rocket, with the robotic Cygnus spacecraft onboard, suffered a catastrophic anomaly moments after launch at NASA's Wallops Flight Facility in Virginia. (Image: © NASA/Joel Kowsky)

Previous analyses & reviews Two independent testing campaigns at Sea Level (more severe) and at High Altitude (6340 ft) (more benign) **System-specific SAR** Measured SL fire environments (3000 K, 2 MW/m²) exceed iridium clad and plutonia melting points Tier I, II, or III Mars Science Laboratory EIS reports risks that exceed NSPM-20 Tier III

thresholds (> 25 rem TED; > 1E-6)

Flight Termination System

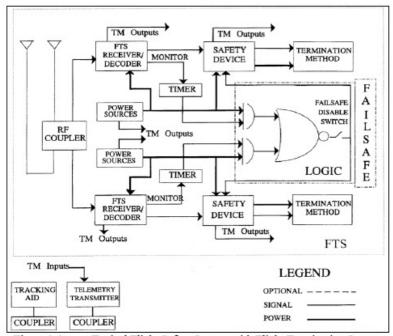


Figure 1-1. Typical Flight Safety System with Flight Termination System RCC 319-14 Flight Termination Systems Commonality Standard

Previous analyses & reviews Assumptions on timing of core booster explosive impulse relative to side boosters drastically affects accident probabilities System-specific SAR FTS custom-designed for Mars **Exploration Rovers and New Horizons** Several different LVs and FTSs Tier I, II, or III

FTS can be designed to favor safety or launch execution

Conclusions

- NSPM-20 likely improves cost & schedule for radioisotope heater units (~32 curies ea.)-only missions like Mars Exploration Rovers and Mars Pathfinder
- NSPM-20 likely does not improve cost & schedule for MMRTG (~60,340 curies ea.) missions like EIM
- Need to consider all aspects of prior safety analysis reports, EISs, Nuclear Risk Assessments, reviews, test results, assumptions/ uncertainties/ unknowns
 - Need to look beyond their face value
- New safety technology can change prior risk results
- NSPM-20 may stagnate safety technology progress

NETS 2021

Thank you for attending!

